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EFFECTS OF MODERATELY HIGH STRAIN RATES ON THE TENSILE PROPERTIES OF METALS

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#### EFFECTS OF MODERATELY HIGH STRAIN RATES ON THE TENSILE PROPERTIES OF METALS

D. P. Moon and J. E. Campbell\*

#### INTRODUCTION

Since the release of DMIC Memorandum 4 on "Effects of High Strain Rates and Rapid Heating on the Tensile Properties of Titanium Allo,s", a number of inquiries have been received by DMIC on the effects of high strain rates on other metals and alloys. Because of the general interest, this memorandum has been prepared to show typical effects of increasing the strain rate to nearly 100 inches per inch per minute at room and alevated temperatures on the tensile properties of a number of commercial alloys. More complete information on specific effects of moderately high strain rates on these and other alloys can be obtained by referring to reports listed in the Bibliography.

Māny investigators have reported on plāstic wave phēnomena during. Exceedingly rapid straining. The loading rates required to produce these phenomena fall into the range of impact tests and are beyond the attempted scope of this memorandum. The works of D. S. Clark\*\*(12-16,20-22) and others are suggested as sources of information on tensile impact properties.

The discussion and data presented in this memorandum also supplement DMIC Report 130 on "Selected Short-Time Tensile and Creep Data Obtained Under Conditions of Rapid Heating".

### EFFECTS OF MODERATELY HIGH STRAIN RATES ON TENSILE PROPERTIES

Increasing the rate of straining during tensile testing is synonymous with increasing the rate of load application. The straining rate and loading rate are proportional up to the elastic limit of the test material. In actual service, it is the rapid application of loads that causes rapid straining. The primary reason for referring to constant strain rates is that, with screw-driven testing machines, relatively constant strain rates are readily obtained with a constant-speed drive mechanism.

In order to show the effects of increasing strain rates, typical tensile-test data are plotted according to stress versus strain rate for tests at room temperature and at elevated temperatures. Plots of typical data for a number of commercial alloys discussed in later sections of this memorandum illustrate the following general trends resulting from increasing the strain rate:

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<sup>\*\*</sup>Numbers in parentheses refer to references in the Bibliography.

- (1) When testing at room temperature, the yield and ultimate tensile strengths of a given alloy either are not affected or are increased slightly as the strain rate is increased.
- (2) When testing at slightly elevated temperatures, the effect of increasing the strain rate is generally similar to that at room temperature, if the alloy has neglegible tendency to creep at stresses lower than the yield strength and remains in a stable condition. However, the yield and ultimate strengths are usually lower, dependent on the temperature. Under these conditions, the length of time at which the specimen is maintained at the testing temperature before it is loaded has little effect on its strength.
- (3) When testing at high températures, increasing the strain rate causes substantial increases in the yield and ultimate strengths, if the alloy tends to creep at stresses lower than the yield strength. As the testing temperature is increased, the yield and ultimate strengths of a given alloy tend to coincide.
- (4) The elastic modulus of a given alloy at a given testing temperature is relatively constant regardless of strain rate.
- (5) Effects of strain rate on ductility are not consistent.
- (6) Reactions which change the structure of the alloy will affect the properties of the alloy when it is heated to a temperature at which the reaction can proceed. These reactions may be precipitation hardening, strain aging, overaging, tempering, recrystallization, and grain growth, or other reactions or combinations of them.

In order to determine how these factors affect the properties of current engineering alloys, a number of testing programs have been initiated. Results of some of these programs are shown in the following sections. Most of this work has been done on sheet specimens with test sections 1/2 inch wide and 2 inches long.

For the data presented in the tables, the specimens that were heated to temperature in 10 or 20 seconds were heated by self-resistance, that is, by passing a controlled electric current through the specimen being tested. In most instances where data for several holding times were available, only data for the shortest holding time (10 seconds) are reported for the rapidly heated specimens. The specimens that were heated to temperature over a period of 15 or 30 minutes were heated by radiation. All strain rates were converted to in./in./min for uniformity in presentation.

#### Aluminum Alloys

Results of tensile tests on three sheet aluminum alloys at strain rates of about 0.003 in./in./min. to 60 in./in./min. and from room temperature to 600 F are shown in Figures 1 to 3. It will be noted that the yield and ultimate strengths at room temperature for the 2014-T6 alloy in Figure 1 were only slightly affected by wide variations in strain rate. For tests made at the standard rate of 0.003 in./in./min, increasing the testing temperature to 500 F resulted in decreasing yield and ultimate strengths to about 12,000 psi. When tested at a strain rate of about 60 in./in./min, however, the yield strength was 30,000 psi at 600 F. Thus a substantial gain in strength was obtained at 600 F under conditions of rapid straining. The curves in Figure 1 also the tendency for the yield and ultimate strengths to occur at nearly the same stress levels under conditions representative of relatively high-temperature behavior for the alloy.

The tensile data for 2024-T3 aluminum alloy in Figure 2 show the effect of precipitation hardening at 450 F during testing. The yield strength at 450 F was higher at a strain rate of 0.003 in./in./min than it was at the higher strain rates. At the low strain rate, there was sufficient time during testing for precipitation hardening to occur before the yield strength was reached. Because of the shorter time required for testing at the higher strain rates, the precipitation-hardening effect on the yield strength was less prominent. At 600 F the specimens had overaged; however, in these specimens, considerably higher strengths were obtained at the higher strain rates.

The effect of increased str in rates for 70.75-T6 sheet aluminum alloy is shown in Figure 3. The same trends are noted for this alloy as for 2014-T6 (Figure 1).

#### Beryllium

Because of its excellent heat-sin; characteristics, its relatively light weight, and its high modulus, beryllium has been considered for certain applications which must withstand aerodynamic heating. Such components are often subjected to rapid straining and rapid heating. Preliminary data as reported by Preston and Kattus(57) on the effect of rapid straining of beryllium bar at room and elevated temperatures are presented in Figure 4.

The room-température tensile strength at the fastest strain rate (20 in./in./min) is relatively low, probably because of the brittleness of the material. At this strain rate, the room-temperature strength is about the same as the strength at 1500 F. At 1000 and 1500 F the specimens had 4.5 and 3.0 per cent elongation, respectively, at the highest strain rate. The properties of beryllium appear to be more attractive at elevated temperatures than at room temperature.

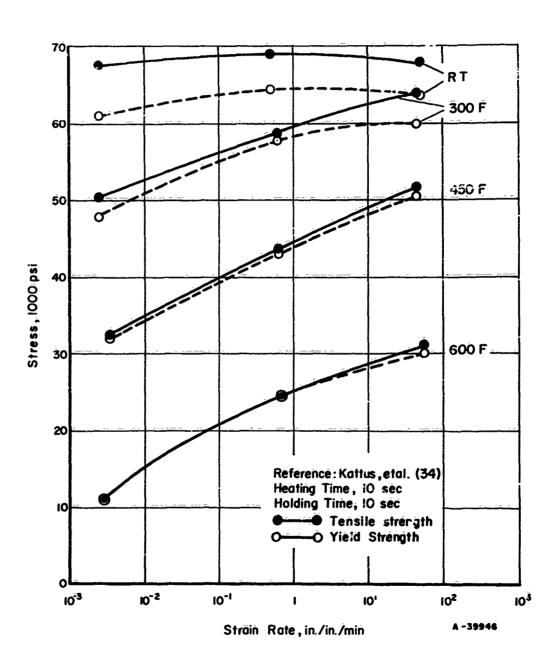


FIGURE 1. ALCLAD 2014-T6 ALUMINUM ALLOY SHEET, TRANSVERSE SPECIMENS, 0.064-INCH THICK

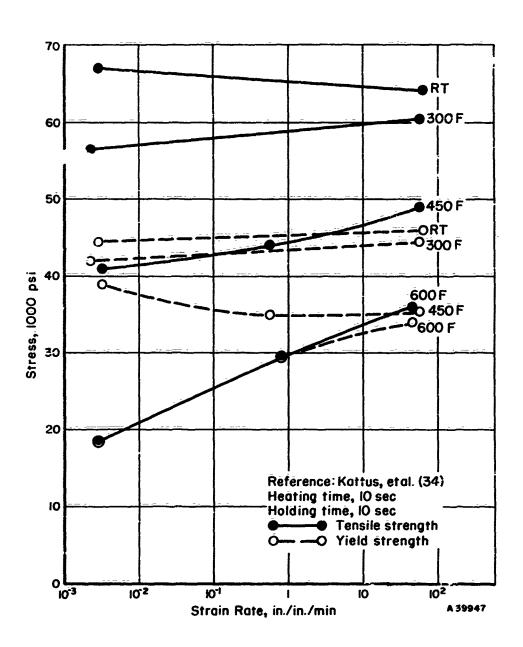


FIGURE 2. ALCLAD 2024-T3 ALUMINUM ALLOY SHEET, TRANSVERSE SPECIMENS, ^.054-INCH THICK

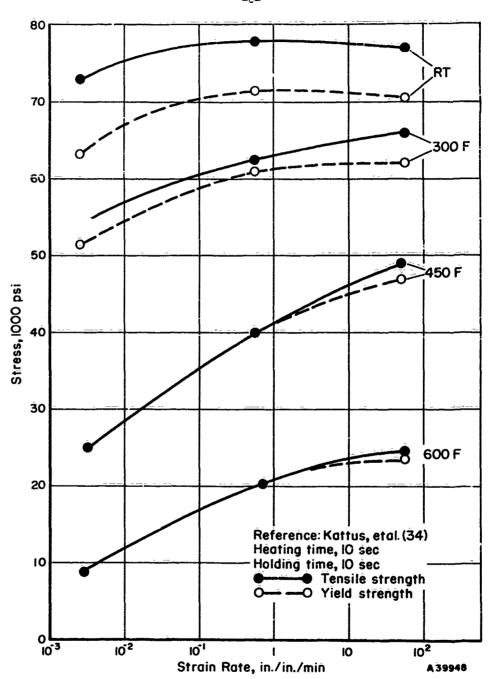


FIGURE 3. ALCLAD 7075-T6 ALUMINUM ALLOY SHEET, TRANSVERSE SPECIMENS, 0.064-INCH THICK

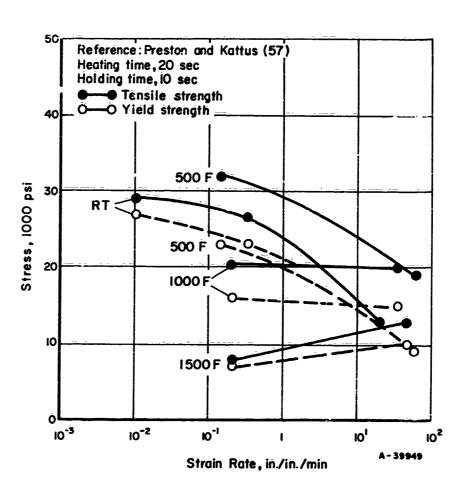


FIGURE 4. UNALLOYED BERYLLIUM BAR TESTED IN ARGON

#### Low Alloy Steel

Results of tensile tests over a wide range of strain rates are shown for normalized AISI 4130 steel sheet in Figure 5 and for quenched—and—tempered 4130 steel sheet in Figure 6. The general trends for effects of increasing the strain rate at room and elevated temperatures, as discussed previously, are valid for this and similar steels.

However, when the testing temperature exceeds the tempering temperature, the strength will be decreased as a result of the tempering effect. This reaction is dependent both on temperature and on time, but for this steel, varying the holding times from 10 to 1,800 seconds had little effect on the strengths at each of the strain rates. Therefore, the tempering effect may arrow within the shortest holding time (10 seconds). Furthermore, for the 4.30 steel, the increase in strength with increased strain rates at 1200 F is primarily an effect of the strain rate, per se. When the tempering effect occurs at a slower rate than in the 4130 steel, rapid heating and straining could occur before the material becomes stabilized (when the maximum temperature is above the tempering temperature). In this case, higher strengths would be obtained.

#### Hot-Work Die Steels

Because of the high strengths that can be achieved, hot work die steels are being considered for a number of highly stressed airborne components. These steels also retain considerable strength at elevated temperatures. The data plotted in Figures 7 and 8 are for two of the hot-work die steels tempered at 950 F. It will be noted that yield strengths of about 150,000 psi were obtained at 1200 F at strain rates of about 30 in./in./min with holding times of 10 seconds. Holding times of 30 minutes at 1200 F resulted in considerably lower strengths than with the short holding times. Apparently, the tempering reaction for these steels requires considerable time at 1200 F. Therefore, if service temperatures exceed the tempering temperature but are attained rapidly and if the component is loaded rapidly, it will withstand considerably higher stresses for a short period of time than would be possible with longer heating times.

#### Austenitic and Heat-Treatable Stainless Steels

Results of tensile tests over a wide range of strain rates for representative stainless steels are shown in Figures 9 to 13. The plotted data in Figure 9 show that annealed Type 301 stainless steel is not strain-rate sensitive until a testing temperature of about 1500 F is reached. At this temperature, the ultimate strength is increased as the strain rate is increased, although the yield strength is not affected by the strain rate.

The curves in Figure 10 are for full-hard Type 301 stainless steel. Data for the full-hard alloy were not obtained at the same temperatures as for the annealed alloy, but a comparison of the data in Figures 9 and 10 will

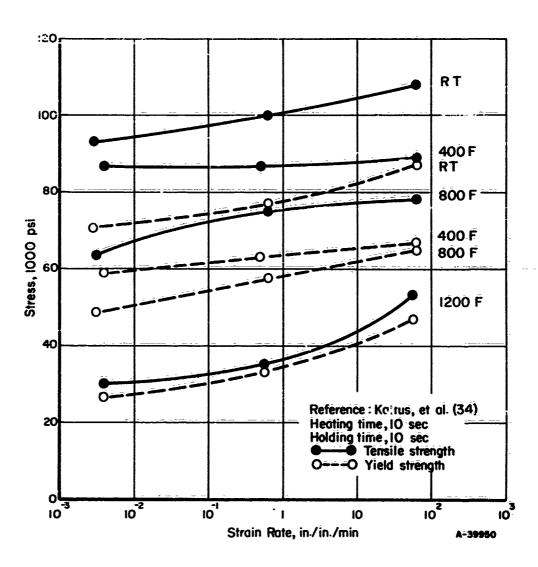
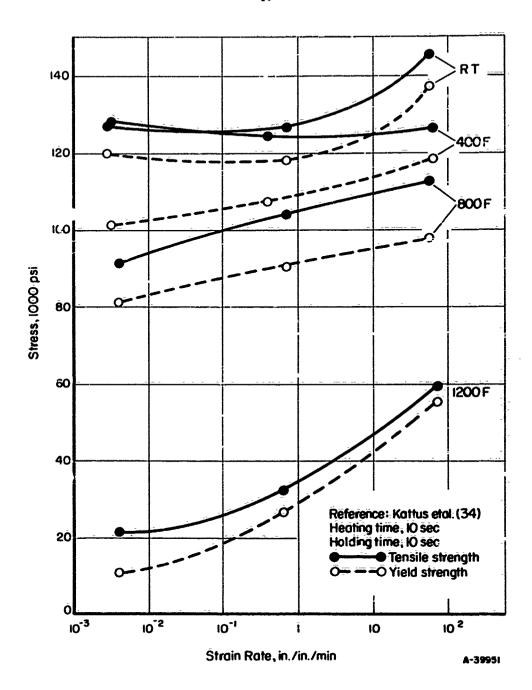


FIGURE 5. NORMALIZED AISI 4130 ALLOY STEEL SHEET, LONGITUDINAL SPECIMENS, 0.040-INCH THICK



FIĞURE 6. HARDENED AISI 4130 STEEL SHEET, LONGITUDINAL SPĒCĪMĒNS, 0.040-INCH THICK (Aŭstēnitizēd 1570 F, oil quenched, tempērēd at 1000 F)

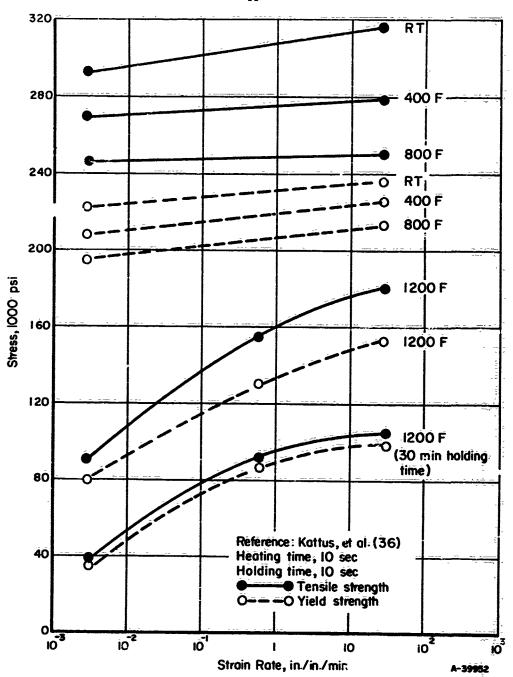


FIGURE 7. HEAT-TREATED THERMÔLD J DIE ŜTEEL SHEET, LÔNGITUDINAL ŚPĒCIMENS, 0.050-ÎNĞH THIĞK (Austenitizêd 1850 F 15 min, oil quênched, double têmperêd 950 F 2 hr)

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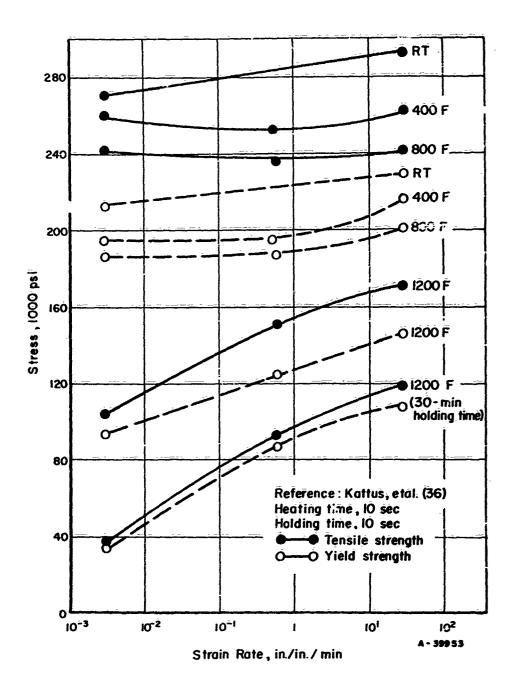
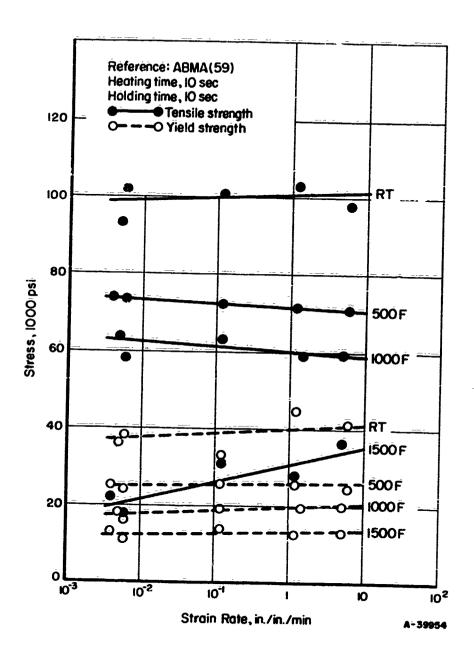


FIGURE 3. HEAT-TREATED PEERLESS-56 DIE STEEL SHEET, LONGITUDINAL SPECIMENS, 0.050-INCH THICK (Austenitized 1950 F 15 min, oil quenched, double tempered 950 F 2 hr)



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FIGURE 9. ANNEALED TYPE 301 STAINLESS STEEL SHEET, LONGITUDINAL SPECIMENS

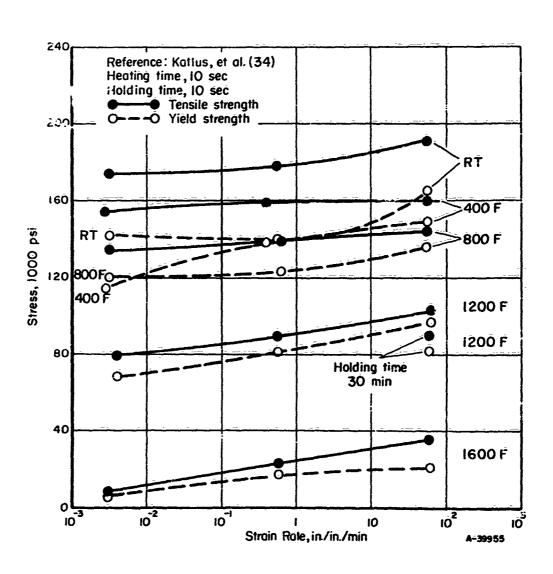


FIGURE 10. FULL-HARD TYPE 301 STAINLESS STEEL SHEET, LONGITUDINAL SPECIMENS, 0.040-INCH THICK

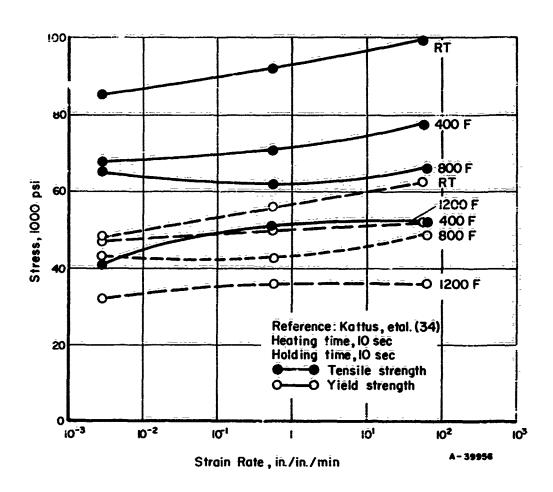


FIGURE 11. \*NNEALED AND DULL CÓLĎ-ROLLEĎ TÝPE 321 STAINLËSS STEEL SHÉET, LONGITUDINAL SPECIMĖNS, 0.040-INĆH THICK

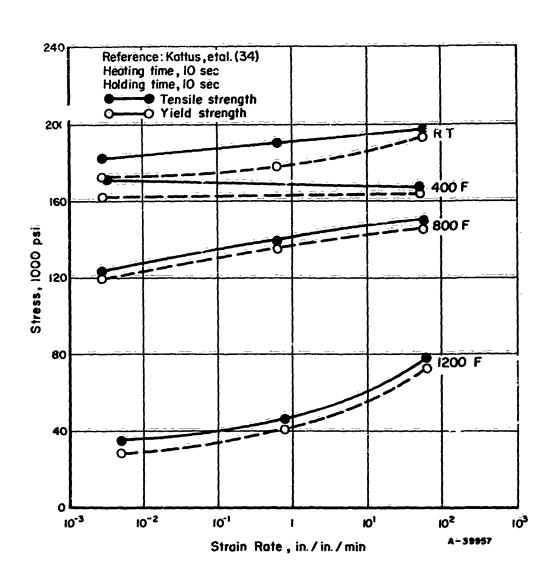


FIGURE 12. HARDENED 17-7 PH STAINLESS STEEL SHEET, TRANSVERSE SPECIMENS, 0.040-INCH THICK (1400 F 90 min, air cooled to RT and water quenched to 60 F, 1050 F 90 min, air cooled)

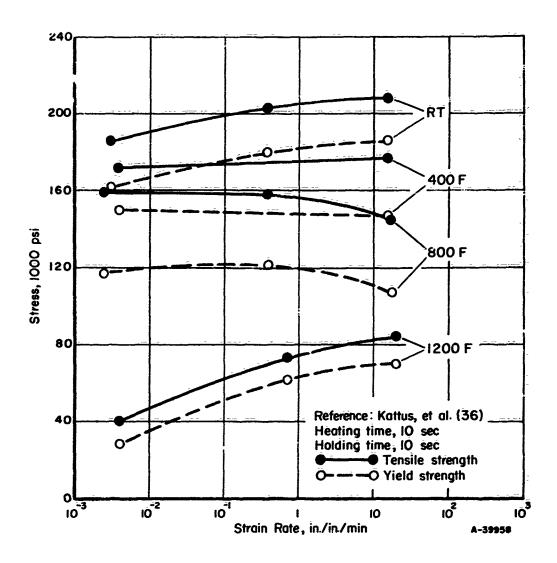


FIGURE 13. HARDENED AM-350 STAINLESS STEEL SHEET, LONGITUDINAL SPECIMENS, 0.040-INCH THICK (1725 F 30 min, water quenched, -100 F 3 hr, 850 F 3 hr)

show the marked increase in strength to 1200 F resulting from cold working. Strengths of the full-hard alloy tended to increase as the strain rate increased. At 1200 F, the strengths were about the same for specimens tested at the slowest strain rate after holding at temperature for 10 seconds and 30 minutes. Tests at the fastest strain rate, however, resulted in lower strengths for the specimens held 30 minutes at temperature. Apparently this is characteristic of the material. On heating to 1600 F, recrystallization and grain growth occur, causing considerably lower strengths. This reaction apparently occurs during heating and within the 10-second holding time, because corresponding specimens that were held at temperature for 30 minutes had about the same strengths.

The data plotted in Figure 11 show the effect of increasing strain rates on the strength of annealed and dull rolled Type 321 stainless steel shows the greatest increase in strength with increasing strain rates. Because of the thermal stability of this alloy, longer holding times at the elevated temperatures had no effect on the tensile properties.

Effects of increasing the strain rate when testing specimens of 17-7 PH stainless steel in the TH1050 condition are shown in Figure 12. Increase in strength with increase in the strain rate was noted for all temperatures except 400 F. The marked reduction in strength from the 800 to the 1200 F testing temperature is the result of the overtempering of these specimens at 1200 F.

Data for tensile tests on AM=350 stainless steel sheet are plotted in Figure 13 for three strain rates. With increased strain rates there was an increase in the room-temperature strength, a decrease in the 800 F strength, and an increase in the 1200 F strength. This steel was overaged at the 1200 F testing temperature.

#### Superalloys

The effects of increased strain rates on the yield strength of N-155 alloy sheet are shown in Figure 14. The effects of increased strain rates from room temperature to 1600 F are only nominal. At 1600 and 2000 F, the yield strengths tend to increase with increased strain rates. These specimens were held at temperature for 15 minutes before testing.

Values for tensile strength at increased strain rates for Unitemp 1753 bar in the solution-treated condition are shown in Figure 15. The marked decrease in strength at testing temperatures above 1450 F is noted in the figure. However, at testing temperatures of 1450, 1600, and 1750 F, the strengths were substantially increased by increasing the strain rate. At temperatures near the solution-treating temperature, increasing the strain rate had less effect on the strength. These specimens were heated to temperature over a 30-minute period and held at temperature for 30 minutes before testing.

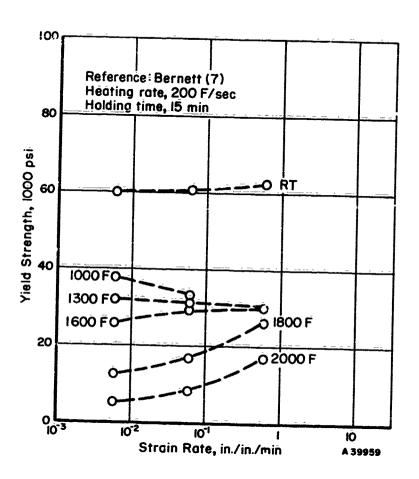


FIGURE 14. N-155 ALLOY SHEET MILL ANNEALED, LONGITUDINAL SPECIMENS, 0.063-INCH THICK

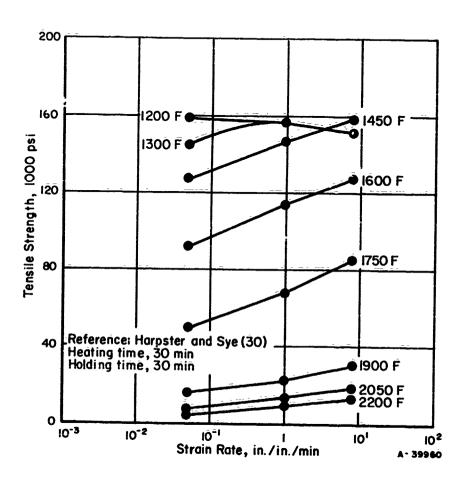


FIGURE 15. UNITEMP 1753 BAR SOLUTION TREATED (2150 F 4 hr, air cocled)

Increasing the strain rate had little effect on the tensile properties of age hardened Inconel X sheet to 1200 F as shown in Figure 16. The tests made at 400 F showed a slight increase in strength with increased strain rates. However, this alloy is relatively stable to 1200 F. The tensile strength at 1600 F increased substantially with increased strain rates.

For Haynes Alloy 25 sheet, Figure 17, the effect of strain rate was most pronounced at room temperature and at 1600 F and above. In the room-temperature tests, the yield strength increased and the ultimate strength decreased at the highest strain rate. Increased holding times at testing temperature had little effect on the tensile properties because of the stability of this alloy.

#### Titanium Alloys

The effects of increasing the strain rate on the tensile properties of 5Al-2.5Sn titanium alloy in the annealed condition are shown in Figure 18. For this alloy, the yield and tensile strengths tend to increase as the strain rate is increased (except the tensile strength of the specimens tested at 400 F at the highest strain rate). This is a stable alpha alloy, and longer holding times had little effect on the properties.

Tensile properties of heat-treated 6Al-4V titanium alloy over a range of strain rates are shown in Figure 19. For this alloy, the strength tended to increase as the strain rate was increased over part or all of the strain-rate range. This alloy is relatively stable to the aging temperature.

Figure 20 illustrates the properties of annealed Ti-8Mn (alpha-beta) alloy sheet at increasing strain rates. It will be noted that the tensile properties were considerably lower at 1000 F than at 600 F. However, at 1000 F, increasing the strain rate resulted in a substantial increase in the tensile properties.

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Data on the effect of increased strain rates on the beta titanium alloy, Ti-13V-11Cr-3Al, were available for the alloy in the annealed condition. These data are plotted in Figure 21. At room temperature, the strengths increase as the strain rate is increased. At 500 and 800 F the trend is reversed. The large decrease in tensile properties at normal testing speeds when increasing the testing temperature from 800 to 1200 F is shown in the figure. When the strain rates are increased for specimens tested in the range from 1200 to 1600 F, the tensile properties are increased substantially.

The marked increase in strength for testing temperatures from room temperature to 1200 F for the aged alloy, as compared to the annealed alloy, is also shown in Figure 21 for a strain rate of about 0.6 in./in./min. Data for other strain rates were not given. However, it appears that there would be little effect of strain rate on the strength to the aging temperature (900  $\tilde{F}$ ). It is expected that tests at 1200 F would show increased strengths with increasing strain rates. There is a marked drop in strength of the aged alloy when increasing the testing temperature from 1200 F to 1275 F. At 1500 F, the annealed and aged alloys have about the same strengths.

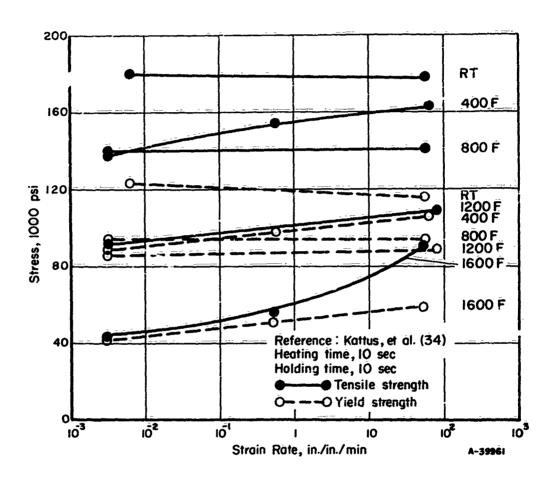


FIGURE 16. AGE-HARDENED INCONEL X SHEET, LONGITUDINAL SPECIMENS, 0.040-INCH THICK (Solution annealed then aged at 1300 F 20 hr, air cooled)

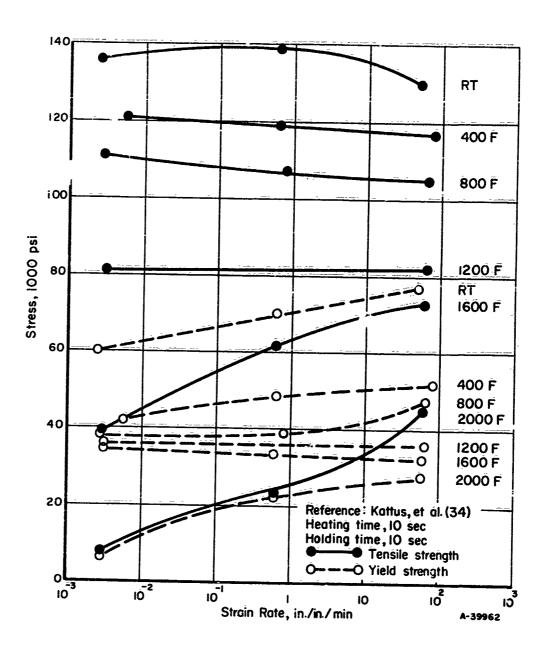


FIGURE 17. ANNEALED HAYNES ALLOY 25 (L-605) SHEET, TRANSVERSE SPECIMENS, 0.040-INCH THICK

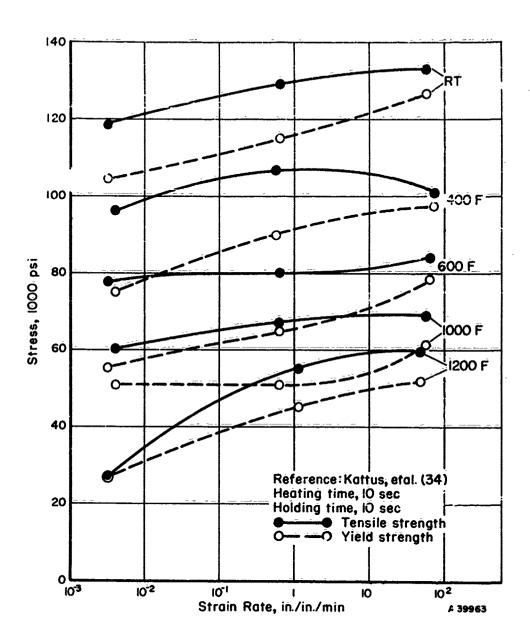


FIGURE 18. ANNEALED Ti-5A1-2.5Sn ALLOY SHEET, LONGITUDINAL SPECIMENS, 0.040-INCH THICK

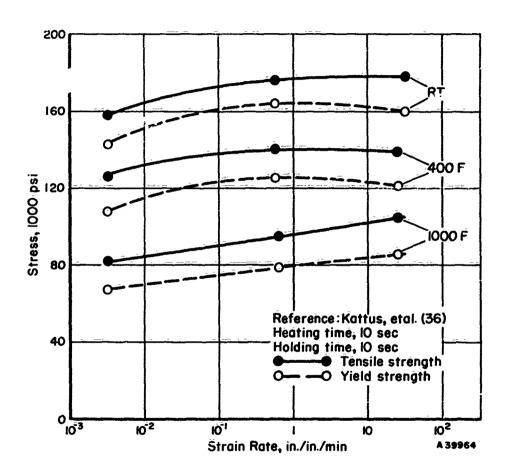


FIGURE 19. AGE-HARDENED TI-6A1-4V ALLOY SHEET, LONGITUDINAL SPECIMENS, 0.040-INCH THICK (1750 F 10 min, water quenched, aged 1000 F 2 hr, air cooled)

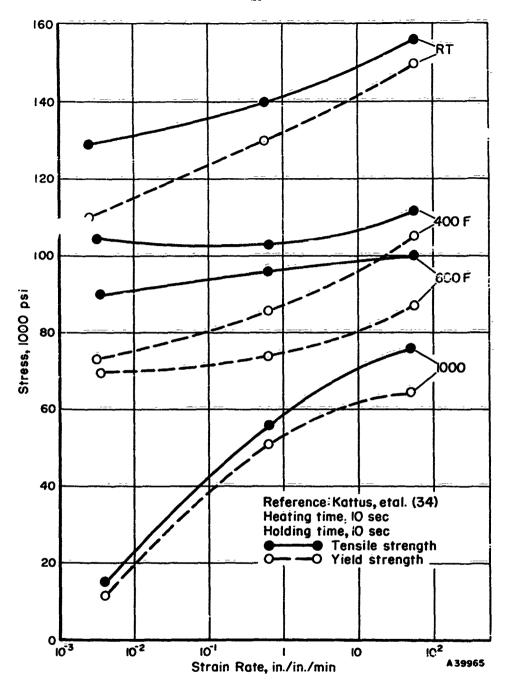


FIGURE 20. ANNEALED Ti-8Mm ALLOY SHEET, LONGITUDINAL SPECIMENS, 0.040-INCH THICK

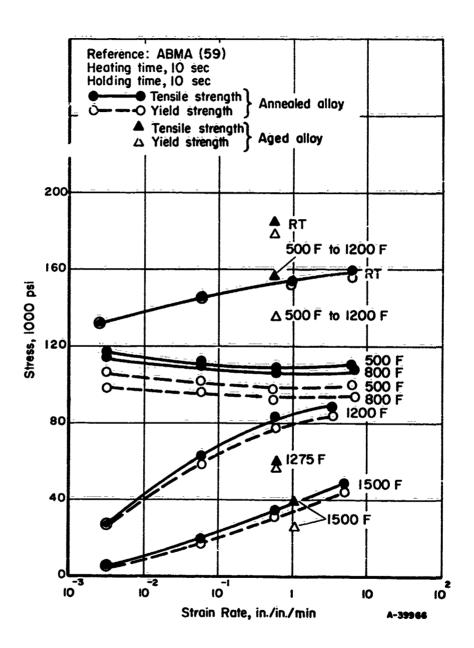


FIGURE 21. ANNEALED AND AGED 13V-11Cr-3A1 TITANIUM ALLOY SHEET
(Aging Treatment: 900 F 50 hr)

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